## MATH3290 Mathematical Modeling 2023/2024 Assignment 3 Due: 5pm, April 16th

Note: Submit your assignment via Blackboard. Late submissions are not allowed.

 Consider the following economic model: Let P be the price of a single item on the market. Let Q be the quantity of the item available on the market. Both P and Q are functions of time. If we consider price and quantity as two interacting species, the following model might be proposed as follows

$$\frac{dP}{dt} = aP(b/Q - P),$$
  
$$\frac{dQ}{dt} = cQ(fP - Q),$$

where a, b, c and f are positive constants.

- (a) Find the equilibrium points of this system in terms of the constants a, b, c and f.
- (b) If a = 1, b = 20,000, c = 1 and f = 30, calculate the equilibrium points of this system using the result of (a).
- (c) Perform a graphical stability analysis to determine what will happen to the levels of P and Q as time increase. Also, classify each equilibrium point with respect to its stability, if possible. If a point cannot be readily classified, explain the reason.
- 2. The fish and game department in a certain state is planning to issue hunting permits to control the deer population (one deer per permit). It is known that if the deer population falls below a certain level m, the deer will become extinct. It is also known that if the deer population rises above the carrying capacity M, the population will decrease back to M through disease and malnutrition. Assume that P is the population of the deer and r is a positive constant of proportionality. The model can be formulated as follows

$$\frac{dP}{dt} = rP(M-P)(P-m).$$

- (a) Write down the explicit formula for the population P in terms of r, m, M and the integral constant (if necessary).
- (b) Show that if P > M for all t, then we have

$$\lim_{t \to \infty} P(t) = M$$

(c) What happens if P < M for all t?

- (d) What are the equilibrium points of the model? Explain the dependence of the steady-state value of P on the initial values of P.
- 3. Consider launching a satellite into orbit using a single-stage rocket. The rocket is continuously losing mass, which is being propelled away from it at significant speeds. We are interested in predicting the maximum speed the rocket can attain.
  - (a) Assume the rocket of mass m is moving with speed v. In a small increment of time  $\Delta t$  it loses a small mass  $\Delta m_p$ , which leaves the rocket with speed u in a direction opposite to v. Here,  $\Delta m_p$  is the small propellant mass. The resulting speed of the rocket is  $v + \Delta v$ . Neglect all external forces (gravity, atmospheric drag, etc.) and assume Newton's second law of motion:

force = 
$$\frac{d}{dt}$$
(momentum of system)

where momentum is mass times velocity. Derive the model

$$\frac{dv}{dt} = \left(\frac{-c}{m}\right)\frac{dm}{dt}$$

where c = u + v is the relative exhaust speed (the speed of the burnt gases relative to the rocket).

(b) Assume that initially, at time t = 0, the velocity v = 0 and the mass of the rocket is m = M + P, where P is the mass of the payload satellite and  $M = \varepsilon M + (1 - \varepsilon)M$  ( $0 < \varepsilon < 1$ ) is the initial fuel mass  $\varepsilon M$  plus the mass  $(1 - \varepsilon)M$  of the rocket casings and instruments. Solve the model in part(i) to obtain the speed

$$v = -c \ln\left[\frac{m}{M+P}\right]$$

(c) Show that when all fuel is burned, the speed of the rocket is given by

$$v_f = -c \ln\left[1 - \frac{\varepsilon}{1+\beta}\right]$$

where  $\beta = P/M$  is the ratio of the payload mass to the rocket mass.

(d) Find  $v_f$  if c = 3 km/sec,  $\varepsilon = 0.8$  and  $\beta = 0.01$ . (These are typical values in satellite launchings.)